

1 Direct numerical simulation of transitional pulsatile stenotic flow using 2 Lattice Boltzmann Method

3 Kartik Jain^{1, 2, a)}

4 ¹⁾ *Simulation Techniques and Scientific Computing, University of Siegen, Hölderlinstr. 3, 57076 Siegen,*
5 *GERMANY*

6 ²⁾ *Center for Biomedical Computing, Simula Research Laboratory, N-1325 Lysaker,*
7 *NORWAY*

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The present contribution reports direct numerical simulations of pulsatile flow through a 75% eccentric stenosis using the Lattice Boltzmann Method (LBM). The stenosis was previously studied by Varghese, Frankel, and Fischer¹ in a benchmark computation, and the goal of this work is to evaluate the LBM and the solver *Musubi* for transitional flows in anatomically realistic geometries. A part of the study compares the LBM simulation results against the benchmark and evaluates the efficacy of most basic LBM scheme for simulation of such flows. The novelty lies in the computation of Kolmogorov micro-scales by performing simulations that consist of up to $\sim 700 \times 10^6$ cells. Recommendations on the choice of spatial and temporal resolutions for simulation of transitional flows in complex geometries naturally arise from the results.

The LBM results show an excellent agreement with the previously published results thereby validating the method and the solver *Musubi* for the simulation of transitional flows. The study suggests that with a prudent calibration of the parameters, the LB method, due to its simplicity and compute efficiency has advantages for the simulation of such flows.

9 I. INTRODUCTION

10 Direct numerical simulation (DNS) is a way of
11 numerically simulating flow in arbitrary geometries
12 by resolving all the temporal and spatial scales that
13 might appear in a transitional or a turbulent flow.
14 Consequently this technique requires very high spatial
15 and temporal resolutions and more compute
16 power. Spectral methods and classical computational
17 fluid dynamic (CFD) techniques like finite
18 element method (FEM) and finite volume method
19 (FVM) have been commonly employed for the simulation
20 of flows. Spectral methods indeed are the
21 most well established technique for the simulation
22 of transitional *incompressible* flows as they allow for
23 an increase in effective resolution with ease. In complex
24 anatomical geometries however, which are the
25 main goal of this and related work² are still difficult
26 to be computed using spectral methods.

27 The Lattice Boltzmann Method (LBM) is an alternative
28 technique for the simulation of low Mach
29 number incompressible flows³⁻⁶. Although well established,
30 due to its novelty the LBM results are
31 sometimes met with skepticism, much of which is
32 attributed to its *indirect* nature i.e. the method converges
33 to the incompressible Navier-Stokes equations
34 under the continuum limits of low Mach and Knudsen
35 numbers^{4,7} instead of a direct discretization of
36 the Navier-Stokes equations. A question then materializes
37 if DNS from such an indirect method is
38 indeed direct. A comparison of LBM with spectral
39 methods by Succi, Benzi, and Higuera³ suggested
40 excellent agreement between the two methods although
41 not much work has been done in this direction
42 after that. A reliable computation of transitional
43 flow with LBM requires an effective tuning of
44 the so called lattice parameters, as the errors in LBM
45 that scale with the order of squared Mach number
46 (Ma^2) can lead to unassuring results⁸. The excellent
47 compute efficiency of the LBM algorithm however
48 makes it a promising method for simulation of
49 transitional flows in complex geometries at large scale

a) Electronic mail: kartik.jain@uni-siegen.de

50 with the advent of modern supercomputers.

51 The LBM solver *Musubi*⁹⁻¹¹ was specifically de- 52 signed for high performance computing architectures 53 to address large scale problems, and it scales on all 54 the federal compute resources of GERMANY namely 55 *Juqueen*, *SuperMUC* and the *Hazel Hen*. *Musubi* 56 solver, even though is verified and validated thor- 57 oughly^{9,10,12} for laminar and turbulent flows, a thor- 58 ough validation has not been done for transitional 59 and pulsating flows¹. Swayed by the need for val- 60 idation, and in support of its extensive use in on- 61 going research efforts for the simulation of transi- 62 tional physiological flows^{2,13}, this work re-simulates 63 the pulsatile flow through the eccentric stenosis that 64 was previously studied in^{1,14}.

65 Since the emphasis is on the LBM, I will partic- 66 ularly focus on the role of parameters like the re- 67 laxation scheme of LBM, and space and time res- 68 olutions in *reproducing* results of previous DNS re- 69 ported in¹. To assess the quality of DNS, I will com- 70 pute and quantify the Kolmogorov length and time 71 scales, and will discuss the conditions under which 72 going down to these scales might benefit the engineer 73 while simulating physiological flows. The results 74 show an excellent agreement with Varghese, Frankel, 75 and Fischer¹ thereby increasing the confidence on 76 the LBM and the solver *Musubi* for such applica- 77 tions. The Kolmogorov micro-scales and the recom- 78 mendations that are provided in this work present a 79 new *state of the art* as no computations, of the or- 80 der of Kolmogorov micro-scales have been reported 81 in literature to the author's knowledge. The results 82 thus have the potential for retrospect in future, and 83 for use as means for comparison.

84 II. METHODS

85 The eccentric stenosis geometry used for this 86 study was similar to the models employed in the ex- 87 periments of stenotic flow by^{15,16}. The stenosis axis 88 was offset by 0.05D, D being the vessel diameter,

¹ *Musubi*, along with other softwares within the APES frame-
work is available as an open source tool for download under:
<https://bitbucket.org/apesteam/musubi>

89 in the eccentric model. The eccentric stenosis ge- 90 ometry used for simulations is shown in figure 1(a). 91 The offset of 0.05D from the axisymmetric counter- 92 part (not studied here) is represented in figure 1(b), 93 where the dashed line shows the eccentric case and 94 black shows the axisymmetric. The presence of ec- 95 centricity here acts as a trigger to transitional flow. 96 The pre and post-stenotic regions of the vessel were 98 respectively extended by 3 and 18 vessel diameters 99 as measured from the throat of stenosis.

100 The Womersley solution for laminar, pulsatile flow 101 through rigid tubes was used as inlet boundary con- 102 dition, which is specified as:

$$\left. \begin{aligned} \frac{u_x}{u_c} &= [1 - r^2] + A \left[1 - \frac{J_0(i^{3/2}\alpha 2r/D)}{J_0(i^{3/2}\alpha)} \right] \sin(\omega t), \\ \frac{u_y}{u_c} &= 0, \\ \frac{u_z}{u_c} &= 0 \end{aligned} \right\} \quad (1)$$

103 where u_c is the cycle-averaged centerline inlet veloc- 104 ity, A is the amplitude of pulsation, J_0 is the Bessel 105 function of type 0, ω is the angular frequency of 106 pulsation, and α is the non-dimensional Womersley 107 parameter ($= \frac{1}{2}D\sqrt{\omega/\nu}$, where ν is the kinematic 108 viscosity). The Womersley parameter defines the ex- 109 tent to which the laminar profile departs from quasi- 110 steadiness, an effect that becomes significant when 111 $\alpha = 3$.

112 The parameters and normalizations mentioned 113 above are chosen to replicate the flow conditions 114 of experiments of Ahmed and Giddens¹⁶ and sim- 115 ulations of Varghese, Frankel, and Fischer¹. The 116 Reynolds number based on the main vessel diam- 117 eter, D, and the mean inlet centerline velocity, u_c was 118 600 with minima and maxima of 200 and 1000. The 119 value of A and α in equation 1 were 0.667 and 7.5 120 respectively. The velocity waveform at the inlet was 121 sinusoidal and recordings were made in intervals of 122 T/6 where T is the period of pulsation (depicted in 123 figure 2).

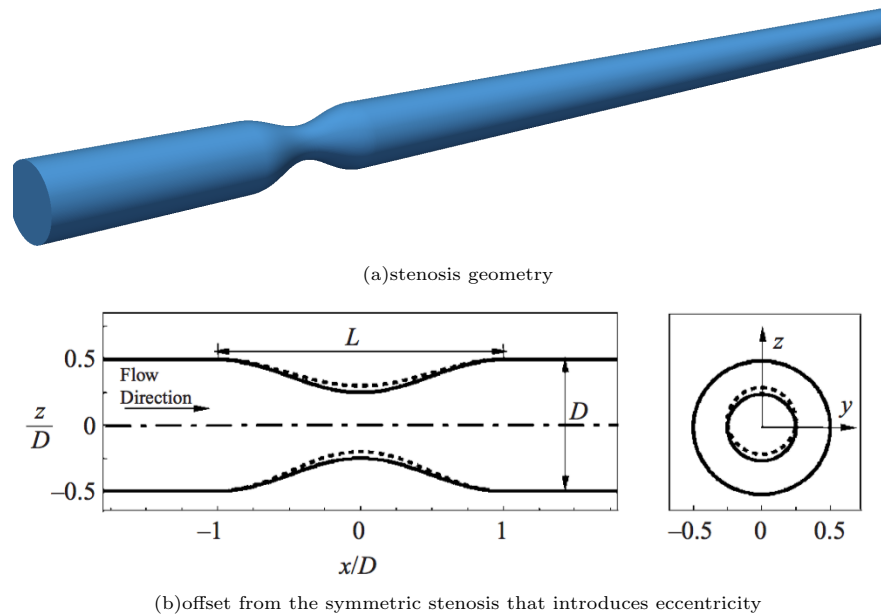


FIG. 1: The eccentric geometry of stenosis used in the study. Lower part of the figure shows front and side views of the stenosis where solid line denotes the axisymmetric model and dashed line denotes the eccentric case. x is the streamwise direction and y and z are cross-stream directions.

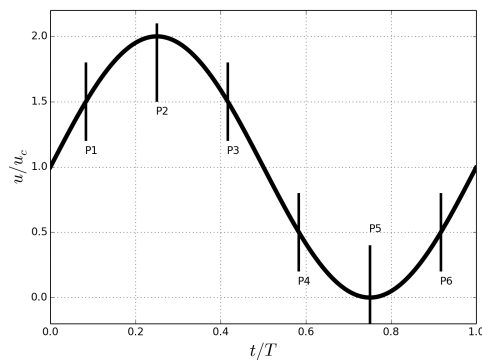


FIG. 2: Axial centerline velocity at the vessel inlet. The measurements were made at 6 time points in the sinusoidal cycle that are indicated in the plot.

125 Direct Numerical Simulations

126 The simulation tool chain is contained in the end-
127 to-end parallel framework APES (adaptable poly en-
128 gineering simulator)^{11,17}. Meshes were created using

	δx	δt	#Cells diameter	#Cells throat	#Cells
LR	64	30×10^{-6}	156	40	$\sim 83 \times 10^6$
HR	32	7.5×10^{-6}	312	80	$\sim 680 \times 10^6$

TABLE I: The spatial and temporal discretization of eccentric stenosis. The space and time have been non-dimensionalized, and of relevance here is the number of cells along the diameter and stenotic throat.

129 the mesh generator *Seeder*¹⁸. I have used the single
130 relaxation scheme Bhatnagar-Gross-Krook (BGK)
131 out of the various LBM relaxation schemes imple-
132 mented in *Musubi* as BGK is the simplest (and
133 most efficient) relaxation scheme of the LBM algo-
134 rithm. I performed two sets of simulations – one
135 with moderate/low resolutions (LR) and one with
136 extremely high resolutions (HR), down to the Kol-
137 mogorov microscales. The resulting parameters are
138 listed in table I. The space and time have been non-
139 dimensionalized for the simulation, and of interest
140 here are the number of lattice cells along the diam-
141 eter of the main channel and that along the throat

143 of the stenosis. The time step is coupled with the
 144 grid spacing in LBM as $\delta t \sim \delta x^2$, which reflects the
 145 *diffusive time scaling* necessary to recover the in-
 146 compressible Navier-Stokes equation from the Lat-
 147 tice Boltzmann Equation⁴. The BGK relaxation pa-
 148 rameter was set to $\Omega = 1.94$ in the present study
 149 that keeps the lattice Mach number within the sta-
 150 bility limits of the LBM^{7,8}. The vessel walls were as-
 151 sumed to be rigid and a no-slip boundary condition,
 152 described by a bounce-back rule in LBM was pre-
 153 scribed. The implementation of this boundary con-
 154 dition ensures stability and provides reasonable ac-
 155 curacy. While other accurate implementations of no-
 156 slip wall approximation could have been employed¹⁹,
 157 this particular boundary condition was chosen to
 158 maintain the principle intention of this study i.e.
 159 employment of *off the shelf* schemes of the LBM to
 160 assess its suitability in such simulations. The D3Q19
 161 stencil of the LBM algorithm was employed which
 162 means 19 discrete velocity directions per fluid cell,
 163 or 19 degrees of freedom. Stencils with larger num-
 164 ber of degrees of freedom can be employed but it has
 165 previously been suggested that the gain in accuracy
 166 for low Re flows is not appreciable compared to the
 167 cost of memory and computation²⁰. At the outlets,
 168 zero pressure was prescribed which is described by
 169 a high-order extrapolation scheme within the LBM
 170 algorithm²¹.

171 LR and HR computations were executed using
 172 1000 and 9600 cores respectively of the *Hazel Hen*
 173 supercomputer installed at the High Performance
 174 Computing center in Stuttgart, GERMANY. The
 175 *Hazel Hen* contains a total of 185 088 cores of In-
 176 tel(R) Xeon(R) CPU E5-2680 v3 (30M Cache, 2.50
 177 GHz). *Hazel Hen* is one of the main federal com-
 178 pute resources in Germany and is ranked at number
 179 8 in the current listing of top supercomputers². A
 180 detailed account of the performance and scalability
 181 of *Musubi* can be found elsewhere^{9,11}. Computation
 182 of each cycle required ~ 36 minutes for LR simu-
 183 lations and ~ 32 minutes for the HR simulations.
 184 The compute time mentioned here seems remark-
 185 ably efficient, but is not comparable with Varghese,
 186 Frankel, and Fischer¹ as those computations were

187 done in 2007 using a completely different architec-
 188 ture and CPUs, and a comparison of computational
 189 efficiency is not the intention of this study.

190 Flow analysis

The analysis of a turbulent or transitional flow follows the statistical principles as statistics, due to the chaotic behavior of the flow are the only reproducible quantities^{22,23}. A total of $n = 22$ (where initial 2 cycles are discarded from analysis) cycles were computed from both LR and HR simulations and were ensemble averaged for analysis. The ensemble average over n cycles is defined as:

$$\bar{u}(x, t) = \frac{1}{n} \sum_{k=0}^{n-1} u(x, t + kT) \quad (2)$$

where $u(x, t)$ is the instantaneous point wise velocity field, \mathbf{x} denotes the spatial coordinates, \mathbf{t} is the time and \mathbf{T} is the period of cycle. The instantaneous three-dimensional velocity field was decomposed into a mean and a fluctuating part using Reynolds' decomposition i.e.

$$u_i(x, t) = \bar{u}_i(x) + u'_i(x, t) \quad (3)$$

The Turbulent Kinetic Energy (TKE) is derived from the fluctuating components of the velocity in 3 directions as:

$$k = \frac{1}{2} (u_x'^2 + u_y'^2 + u_z'^2) \quad (4)$$

191 A power spectral density of the TKE, computed us-
 192 ing Fourier transforms provides information about
 193 the frequency components present in the flow, and
 194 can be related to the Kolmogorov energy decay.

195 The Q-criterion

The Q-criterion was preferred in the present study for the visualization of coherent flow structures as it shares properties with both the vorticity and pressure criterion²⁴. The Q-criterion is the second invariant of the velocity gradient tensor $\nabla \mathbf{u}$, and reads:

$$Q = \frac{1}{2} (\Omega_{ij} \Omega_{ij} - S_{ij} S_{ij}) \quad (5)$$

² <http://top500.org>

where

$$\Omega_{ij} = \frac{1}{2} \left(\frac{\partial u_i}{\partial x_j} - \frac{\partial u_j}{\partial x_i} \right) \quad (6)$$

and

$$S_{ij} = \frac{1}{2} \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) \quad (7)$$

196 are respectively the anti-symmetric and symmetric
197 components of $\nabla \mathbf{u}$.

The Q-criterion can be physically viewed as the balance between the rotation rate $\Omega^2 = \Omega_{ij}\Omega_{ij}$ and the strain rate $S^2 = S_{ij}S_{ij}$. Positive Q isosurfaces confine the areas where the strength of rotation overcomes the strain - making those surfaces eligible as vortex envelopes. Several interpretations of Q-criterion have been proposed, see for example Robinson²⁵ which recasts Q in a form which relates to the vorticity modulus ω :

$$Q = \frac{1}{4}(\omega^2 - 2S_{ij}S_{ij}). \quad (8)$$

198 This implies that the Q is expected to remain positive
199 in the core of the vortex as vorticity increases
200 as the center of the vortex is approached.

201 DNS quality assessment with Kolmogorov microscales

The smallest structures that can exist in a turbulent flow are based on Kolmogorov's theory²². Viscosity dominates and the TKE is dissipated into heat at the Kolmogorov scale²². The Kolmogorov microscales are generally described in terms of the rate of dissipation due to the turbulent kinetic energy, which results in equations containing 4th order terms^{22,23}. The Kolmogorov scales, for simplicity, can also be computed in terms of local friction velocity $u_* = \sqrt{\nu||s||}$ where s_{ij} is the *fluctuating* component of strain rate defined as:

$$s_{ij} = \frac{1}{2} \left(\frac{\partial u'_i}{\partial x_j} + \frac{\partial u'_j}{\partial x_i} \right) \quad (9)$$

and ν is the kinematic viscosity. As the physical quantities have been non-dimensionalized for the

present study, the viscosity ν used for the computation of Kolmogorov micro-scales is the *lattice viscosity* that is formulated on the basis of BGK relaxation parameter Ω as:

$$\nu = \frac{1}{3} \left(\frac{1}{\Omega} - \frac{1}{2} \right) \quad (10)$$

The Kolmogorov length, time and velocity scales are then respectively estimated as:

$$\eta \equiv \nu/u_* \quad (11)$$

$$\tau_\eta \equiv \nu/u_*^2 \quad (12)$$

$$u_\eta \equiv u_* \quad (13)$$

Based on these scales, the quality of the spatial and temporal resolution of a simulation is estimated by computing the ratio of δx and δt against corresponding Kolmogorov scales i.e.

$$l^+ = \frac{u_* \delta x}{\nu} \quad (14)$$

$$t^+ = \frac{u_*^2 \delta t}{\nu} \quad (15)$$

202 Ideally these ratios should be ~ 1 but in practice it
203 has been observed that a l^+ of the order of $\mathcal{O}(10)$
204 is usually enough for the simulation of moderate
205 Reynolds' numbers transitional flows²⁶.

206 III. RESULTS

207 Figure 3(a) and 3(b) depict the axial centerline
208 velocities over the last $n = 6$ cycles obtained from
209 LR and HR simulations respectively. Ensemble averaged
210 counterparts for $n = 20$ cycles are shown in
211 figure 4(a) and 4(b). Whereas the main flow captured
212 by LR and HR simulations is similar, high
213 resolutions seem to capture larger fluctuations particularly
214 in post-stenotic regions ($x=3-5D$), and the differences
215 between LR and HR are mostly visible in
216 time periods when the flow starts to decelerate (time

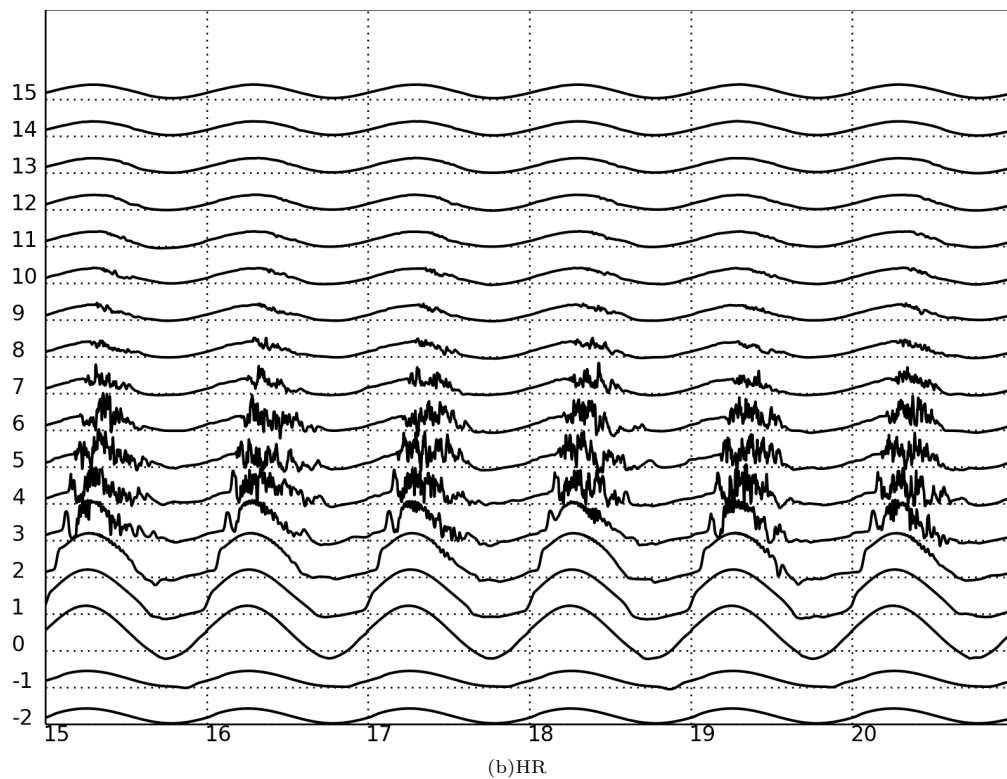
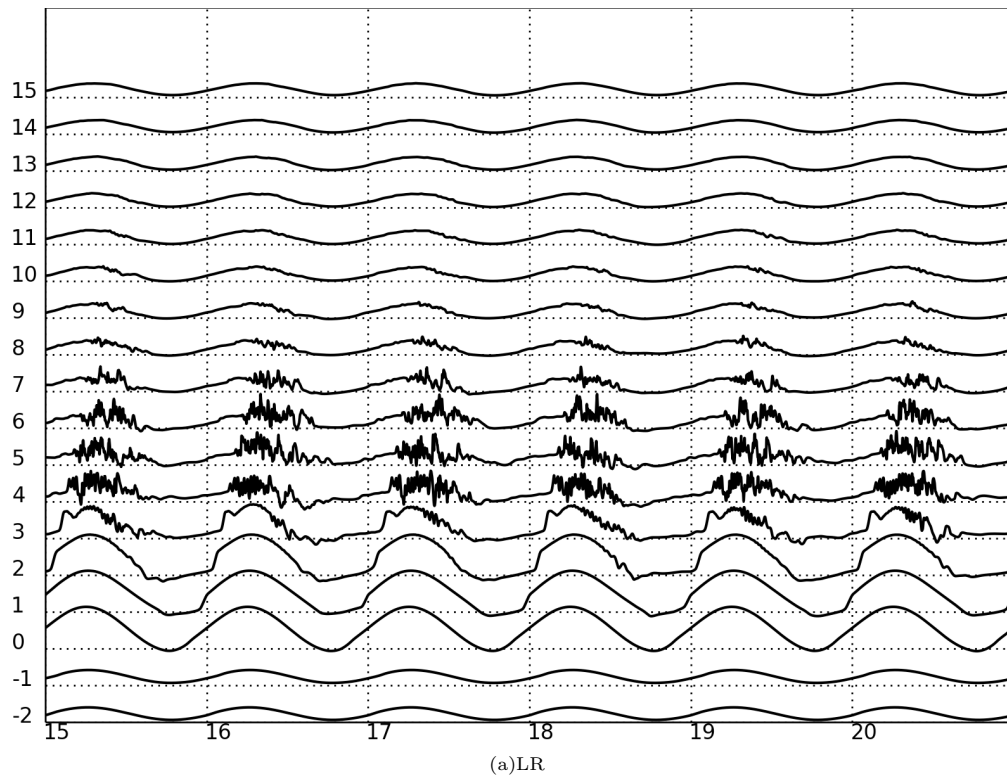


FIG. 3: Temporal evolution of the normalized centerline axial velocity, u/u_c over the last 6 cycles out of total 20 that were simulated, as a function of axial distance through stenosis, shown for LR and HR simulations.

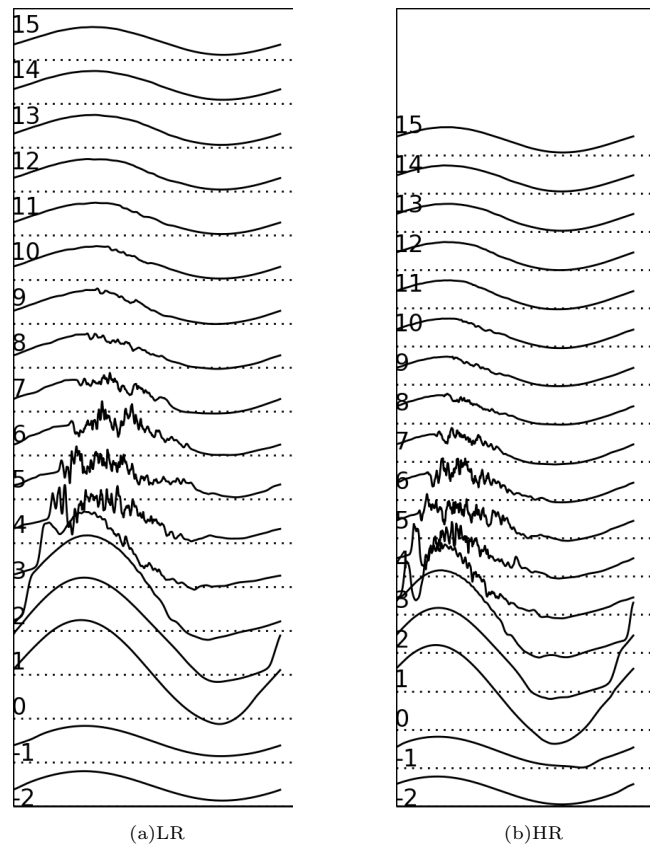


FIG. 4: Normalized centerline axial velocity ensemble averaged for $n = 20$ cycles shown for LR and HR simulations.

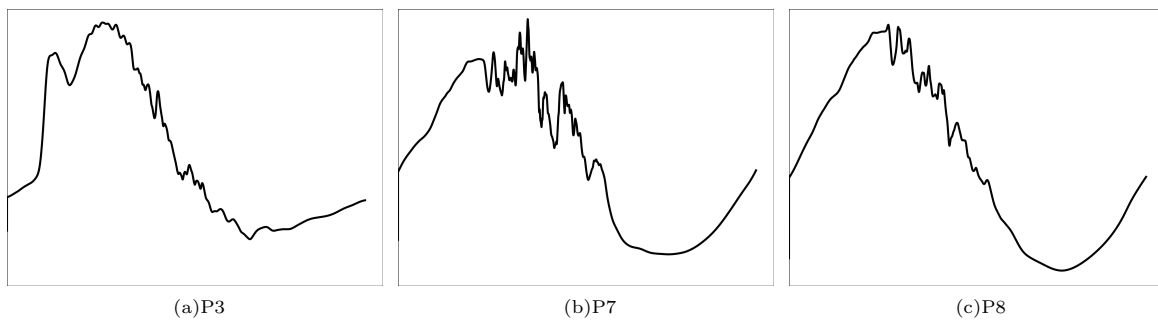


FIG. 5: Ensemble averages at axial centerline locations $x = 3, 7 \& 8D$ to magnify the fluctuations during deceleration and the re-laminarization of flow during acceleration.

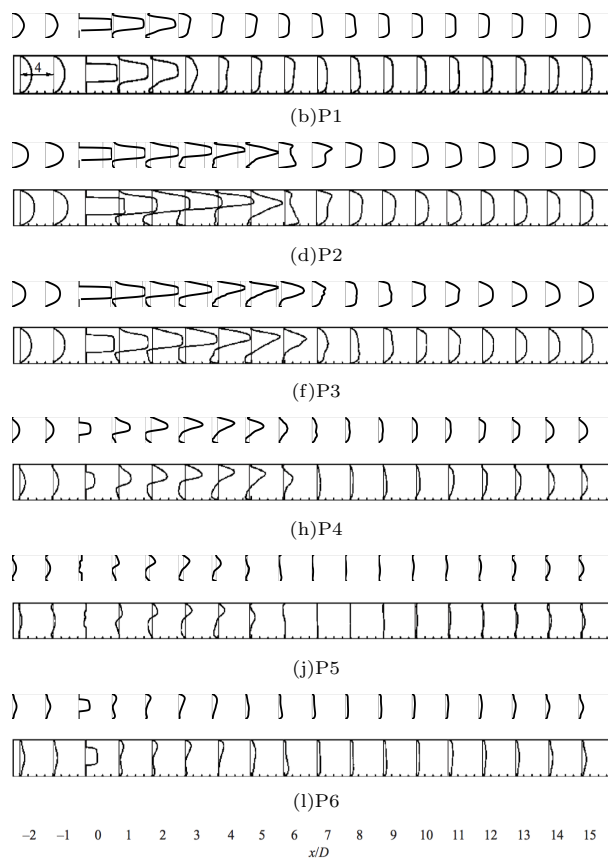


FIG. 6: Sequence of ensemble-averaged axial velocity profiles, $\langle u \rangle / u_c$ at observation points P1-P6 (top down) in the x - z plane. The top row of each point depicts computations from *Musubi* followed by the corresponding image from the benchmark computations from NEK5000.

instants P2 to P4). This observation would characterize LR setup as *converged*, though as would be seen in the turbulent characteristics, some intricate features might be suppressed by low resolutions.

The ensemble averaged quantities look largely similar for LR and HR simulations as the minute dynamics that were captured by HR are smeared out upon averaging. Subtle differences remain in these post-stenotic regions due to higher gradients in these regions. The remainder of the text would thus employ LR simulation results when ensemble averaged quantities are discussed and HR will be talked about only when instantaneous quantities are of interest.

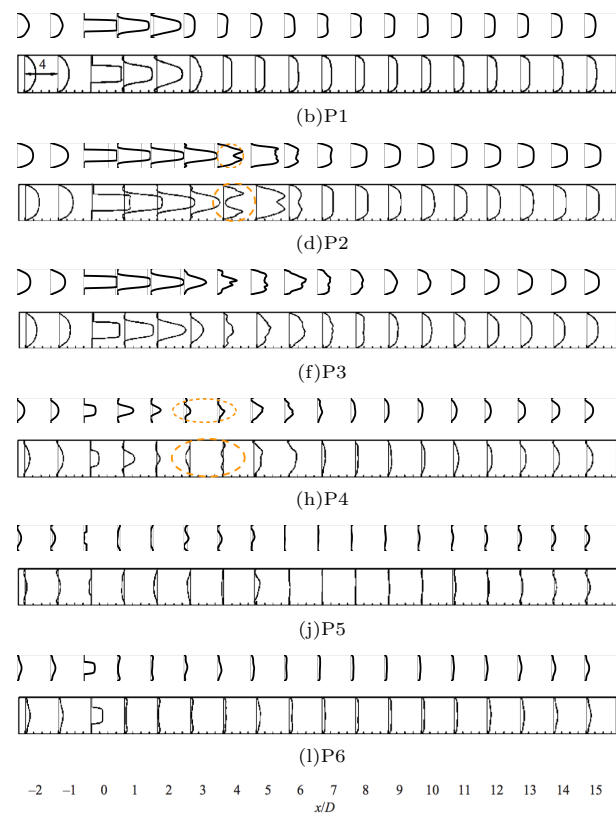


FIG. 7: Sequence of ensemble-averaged axial velocity profiles, $\langle u \rangle / u_c$ at observation points P1-P6 (top down) in the x - y plane. The top row of each point depicts computations from *Musubi* followed by the corresponding image from the benchmark computations from NEK5000.

Figure 5 shows the ensemble average at axial centerline locations $x = 3, 7$ & $8D$ from fig. 4(a) to highlight the loss of coherence patterns in the flow during deceleration phases, and the re-laminarization of the flow during acceleration. The stabilization of the flow in late-acceleration phases is location dependent, as the fluctuations seem to reduce beyond $x > 7D$. The regions between $x = 4D$ and $x = 6D$ represent highly chaotic behavior and larger cycle-to-cycle variations whereas the flow starts to become laminar in regions far-off from the stenosis throat.

Figure 6 and 7 respectively show the upstream velocity field in xz and xy axial bisecting planes. The velocity is ensemble averaged for $n=20$ cycles after

244 2 initial cycles that have been discarded from anal-
 245 ysis. Corresponding plots from Varghese, Frankel,
 246 and Fischer¹ are also shown below each plot com-
 247 puted from *Musubi* for a direct visual comparison.
 248 An overall agreement is portrayed by this figure and
 249 the noticeable differences are highlighted under or-
 250 ange circles, and details are discussed in section IV.

251 Turbulent Characteristics of the Flow

252 Figure 8 shows the frequency spectra of the tur-
 253 bulent kinetic energy computed from centerline ax-
 254 ial velocity at varying distance from the stenosis
 255 throat from LR simulations whereas figure 9 shows
 256 the same from HR simulations. The fluctuations
 259 are higher at locations $x=3-6D$ and the flow starts
 260 to re-laminarize beyond $x=9D$ – an observation that
 261 is consistent with the instantaneous and ensemble
 262 averaged velocity fields. The spectra at these loca-
 263 tions indicate a large number of frequencies in the
 264 inertial subrange. The viscosity starts to dominate
 265 at frequencies of $\sim 10^4$ Hz up to $x=7D$. At locations
 266 $x > 10D$, flow can be considered largely laminar as
 267 the PSD is mostly below 10^{-10} for frequencies more
 268 than 10^3 Hz.

269 Comparison of the power spectrum plots from
 270 LR and HR simulations reveals a generic pattern
 271 whereby higher frequencies are detected by HR sim-
 272 ulations due to small δt . The qualitative patterns are
 273 similar for both resolutions and as discussed later,
 274 the choice of resolutions is generally case dependent.

275 Vortex structures

276 Figure 10 and 11 show the ensemble averaged vor-
 277 ticity magnitude across the xz and xy planes respec-
 278 tively from LR simulations. The main patterns look
 279 similar to Varghese, Frankel, and Fischer¹, though
 280 LBM has detected some miniature vortices due to
 281 higher spatial and temporal magnitudes. It may be
 282 explicitly mentioned that the ensemble average for
 283 $n = 3$ cycles only is taken to educe the vortices,
 284 which would otherwise be smeared out if ensem-
 285 ble average over larger number of cycles are taken.
 286 When ensemble average for larger number of cycles

287 is taken, the vortices are expected to die away and
 288 mimic Varghese, Frankel, and Fischer¹ more closely,
 289 as due to cycle-to-cycle variations, the location of
 290 vortices rapidly changes from one cycle to another.
 291 The eduction of instantaneous miniature vortices is
 292 in fact an important feature of DNS and is represen-
 293 tative of the excellent control of numerical viscosity
 294 in LBM algorithm^{3,13}.

295 Figure 12 shows the vortex structures at 4 obser-
 296 vation points during the 22nd cycle for LR simula-
 297 tions. The vortices are educed by the Q -criterion
 298 discussed in subsection II. The vortices begin to die
 299 during P5 and P6 due to re-laminarization of flow
 300 and are thus not shown in this figure. It is par-
 301 ticularly interesting to observe that the majority of
 302 vortices lie along $x > 4D$ where the flow transits to
 303 main-stream turbulence, and was also reminiscent in
 304 the PSD plots. Some detached vortices are attached
 305 along the stenosis throat which is a consequence of
 306 higher strain along this region.
 307

308 A mentionable aspect of this study is the *cycle-to-*
 309 *cycle* variations in the flow. As a result of transition
 310 to turbulence, the characteristics of flow vary from
 311 one cycle to another, and one cycle is not super-
 312 imposable to another as would be expected, for ex-
 313 ample in a laminar flow. This aspect is highlighted
 314 in figure 13 which shows the centerline velocity at
 315 $x = 3D$ over 10 cycles as thin lines in the back-
 316 ground. The black-line depicts the ensemble average
 317 over the 10 cycles and the dotted black lines rep-
 318 resent standard deviation ($\pm\sigma$). It is immediately
 319 observed that all the lines overlap with each other
 320 during acceleration of the flow as a result of accel-
 321 eration induced re-laminarization whereas there are
 322 large deviations when the flow transitions during de-
 323 celeration phase, and it continues up to the complete
 324 deceleration before re-laminarizing by acceleration.

325 Kolmogorov microscales

326 Table II lists the l^+ and t^+ for LR and HR simu-
 327 lations. The rations are computed at the observation
 328 point P2 along $x=4D$ during the 20th cycle as the
 329 fluctuation in strain rate was maximal at this loca-
 330 tion during P2, which implies maximum dissipation
 331 during the whole simulation. The employed resolu-

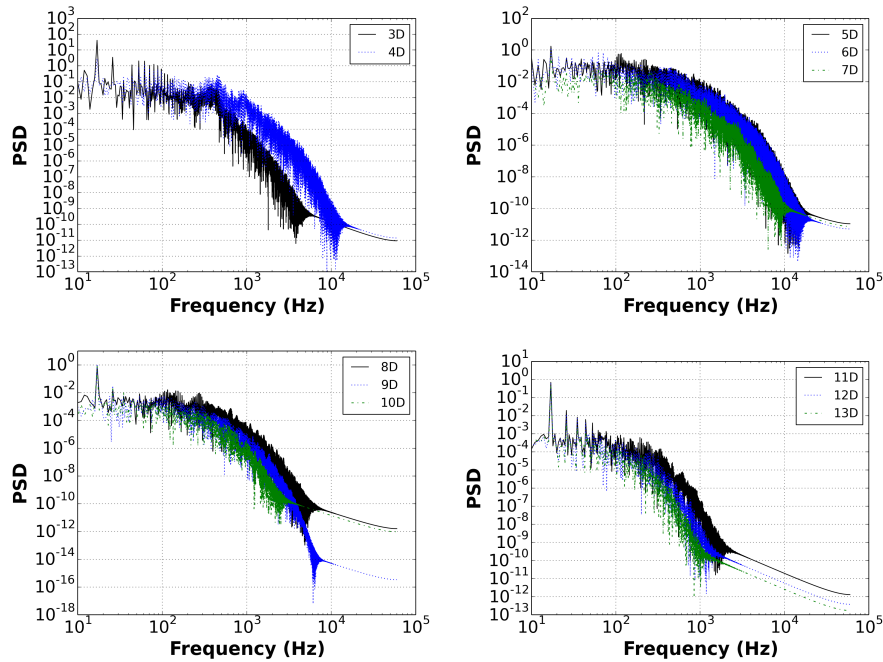


FIG. 8: Energy spectra of the turbulent kinetic energy computed at the centerline from LR simulations. The locations represent distance in diameters from the stenosis.

	δx	δt	l^+	t^+
LR	64	30×10^{-6}	2.67	0.84
HR	32	7.5×10^{-6}	1.10	0.53

TABLE II: The ratio of spatio-temporal scales (l^+ , t^+) in the simulation and the Kolmogorov microscales for different resolutions.

332 tions are ample to resolve the rapidly varying struc-
 333 tures expected in a turbulent flow²⁶. Whereas the δx
 334 of LR is ~ 3 times of the Kolmogorov length scale,
 335 in a minor transitional flow in relatively less com-
 336 plex geometry like the presented stenosis, it should
 337 be enough for simulations as is also evident from the
 338 results.

339 IV. DISCUSSION

340 Analysis of the flow

341 Main things to observe (and to compare
 342 against Varghese, Frankel, and Fischer¹) from the
 343 flow patterns of figure 6 and 7 were flow direction,
 344 peaks and nadirs as well as zones of flow reversal or
 345 recirculation. The flow field in the x-z plane exhib-
 346 ited satisfactory agreement with Varghese, Frankel,
 347 and Fischer¹. The velocity was elevated at the
 348 throat of the stenosis, remained high in post-stenotic
 349 regions before becoming nearly constant near the
 350 end regions of the channel ($x > 11D$).

351 Similar agreement was seen for the x-y plane –
 352 though there were a few locations of disparity. For
 353 example, the post stenotic field at P2 ($x=4$) looked
 354 very different from Varghese, Frankel, and Fischer¹
 355 (highlighted in orange circles). At P4, even the di-
 356 rection of flow disagreed at $x=3$ and $x=4$, which is

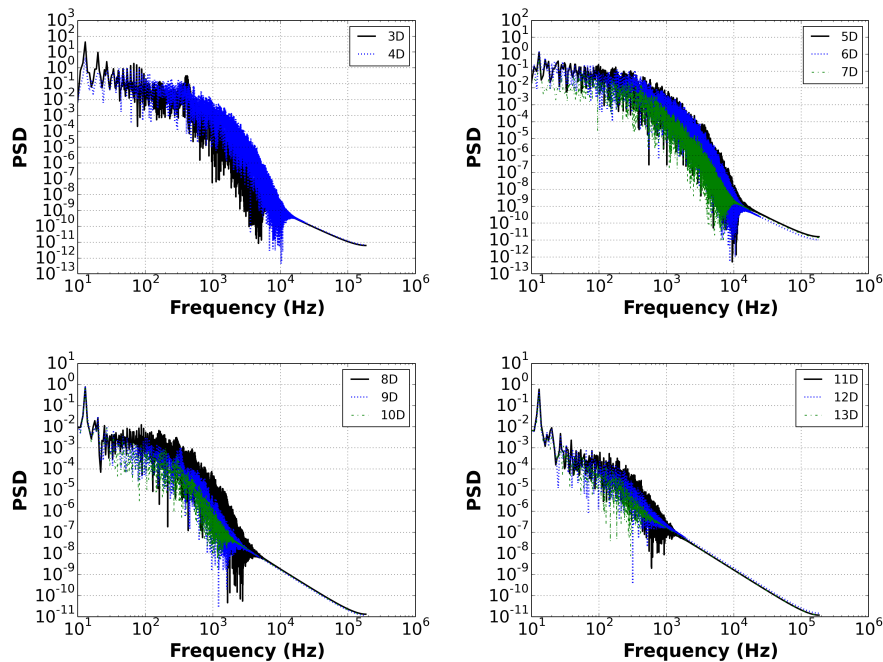


FIG. 9: Energy spectra of the turbulent kinetic energy computed at the centerline from HR simulations. The locations represent distance in diameters from the stenosis.

357 very surprising especially after an excellent agree-
 358 ment that was seen for x-z plane. The exact reason
 359 for this mismatch cannot be stated but it can be at-
 360 tributed to the different solution algorithms where
 361 minor differences are obvious. It should also be
 362 kept in mind that this comparison is one of *statis-*
 363 *tics* and involves round-off errors. Moreover, the
 364 perturbation that might be introduced as a result
 365 of wall roughness may stay in the flow up to an ar-
 366 bitrary number of cycles before being washed out
 367 completely. The boundary layer resolved by LBM
 368 and Spectral methods can be immensely different
 369 due to the distinct algorithms, and accuracy of one
 370 over the other can not be distinctly stated. The
 371 regions that are up-stream of stenosis, and in its
 372 vicinity have very high velocity gradients due to the
 373 onset of transition, and the wall boundary condi-
 374 tions are expected to influence the results dramat-
 375 ically. Whats most contenting is that in spite of
 376 different flow directions in these two locations, the
 377 flow field re-attained similarity beyond $x=5$ at all

378 the time points, as did the flow field right at the
 379 stenosis throat.

380 Also, the flow fields from *Musubi* looked exactly
 381 the same as those from NEK5000 at P6 in the x-
 382 y plane, which portrays that the discrepancies seen
 383 during P4 and P5 might have been a result of the loss
 384 of coherence patterns caused by large decelerative
 385 forces, which were then overcame by the stabiliza-
 386 tion resulted by acceleration of flow. It is expected
 387 that if ensemble averaging over a larger number of
 388 cycles is done, these effects would wash away and
 389 the disparities would eventually vanish. That as-
 390 pect however is not considered important due to the
 391 convincing agreement in other locations and time
 392 points, and would perhaps be useful when the simu-
 393 lation is actually re-conducted with NEK5000 as
 394 well.

395 Particularly interesting was the similarity in the
 396 flow patterns during P2 where the velocity of in-
 397 flow is maximum, and P6 where the flow is in mid-
 398 acceleration phase after the deceleration. The pres-

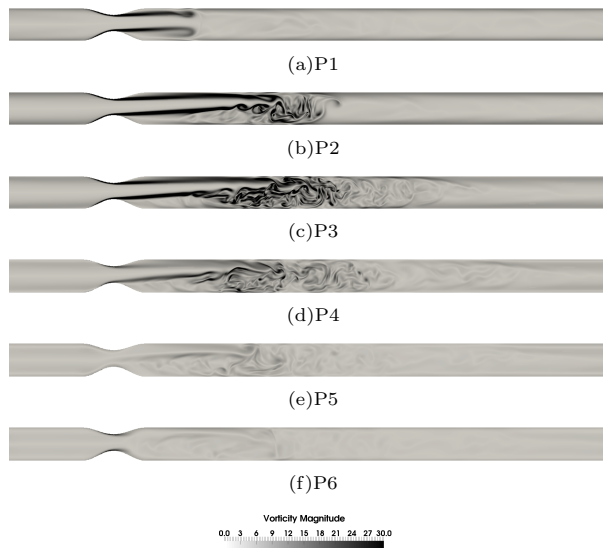


FIG. 10: Ensemble averaged vorticity magnitude, normalized by u_c/D at the x - z plane. Ensemble averaging is performed for ONLY $n = 3$ cycles to observe the intricate vortices which smear out when larger number of cycles are taken into account.

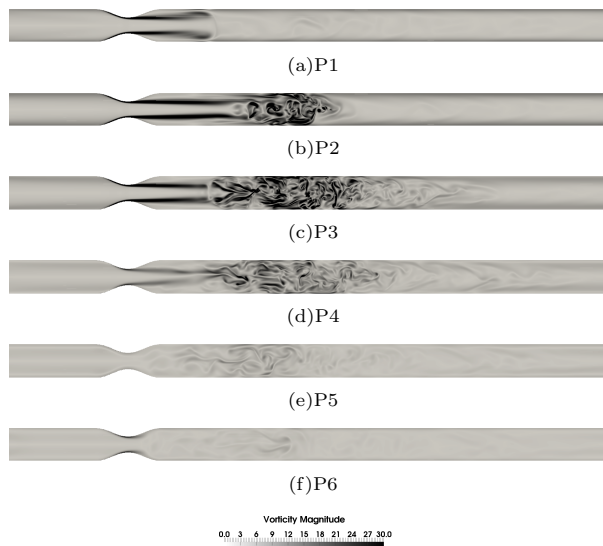


FIG. 11: Ensemble averaged vorticity magnitude, normalized by u_c/D at the x - y plane. Ensemble averaging is performed for ONLY $n = 3$ cycles to observe the intricate vortices which smear out when larger number of cycles are taken into account.

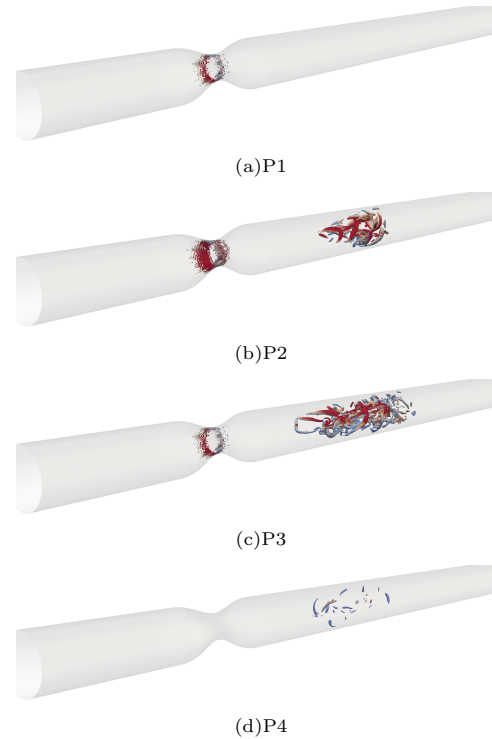


FIG. 12: Velocity colored Q -isosurfaces ($Q=0.4$) at the observation points P1-P4 during the 20th cycle for LR resolution. The velocity is normalized by u_c .

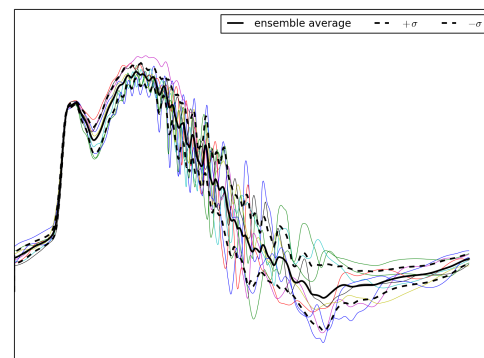


FIG. 13: The axial centerline velocity at $x = 3D$ for 10 different cycles overlapped over each other as thin lines. The black line represents the ensemble average over these 10 cycles whereas the standard deviation ($\pm\sigma$) is depicted by dotted lines.

ence of higher vortices during P3 than P2 as educed by the Q-Criterion (figure 12) appeared surprising at a first glance. From figure 2 one would expect highest vortices during P2 since it is the point with peak velocity. This is a consequence of the large decelerative forces that results in chaotic flow between P2 and P3 and creates distinct and larger vortex envelopes during P3. This was also observed in vorticity plots of figure 10 and figure 11.

The vorticity plots in figure 10 and 11 show some minute vortices in post stenotic areas even after ensemble averaging (though for only $n = 3$ cycles). The overall vorticity patterns are exactly similar to Varghese, Frankel, and Fischer¹. It can easily be seen that if the sharpness of vortices is reduced upon ensemble averaging for more cycles, the patterns will look exactly similar as those in¹. This prognosis is reminiscent of cycle-to-cycle variations that were clearly visible in instantaneous fields of figure 3. The initial conditions to any arbitrary cycle n are fed from the last state of previous cycle $n - 1$, which, in addition to the transitional nature of the flow itself makes flow field of each cycle look different. This is the reason why ensemble averaging is required for the analysis of transitional flows, and as discussed in²⁷, the computation of flow quantities poses additional challenges in such a flow.

Role of employed resolutions

As was seen from figures 3(a) and 3(b), the employment of high resolutions, that was directly at the order of Kolmogorov length and time scales (table II), did not provide much *improvement* to the simulated flow field except for the capture of some rapid spatial and temporal scales while educing larger cycle-to-cycle variations. The consumption of memory and compute time on the other hand became ~ 8 times with a higher resolution, though the computation was still remarkably fast. An unequivocal remark whether LR indeed is enough for simulating transitional flows in general, and transitional physiological flows in particular can not be made because physiological geometries are generally extremely complex. It has been shown in our studies of transitional aneurysmal flows that the solution

indeed does change upon refinements when the geometry is not a regular conduit but a complex geometry^{2,13}. Choice of resolutions in fact has been a subject of discussion with other numerical techniques as well, see for example research on this aspect about blood flow in aneurysms²⁸⁻³⁰ as well as simulation of cerebrospinal fluid in the spinal canals³¹.

In addition to that, the stenosis geometry studied here has one outflow which is perpendicular to the incoming flow. Presence of more outlets as well as the angle at which the downstream flow attacks these outlets is likely to upsurge the resolution requirements. Moreover, the stenosis may be viewed as a *controlled distortion* in a straight cylindrical pipe which is located at only one location in the pipe. In anatomically realistic geometries, arbitrary distortions can be present at multiple locations. The irregularity of such distortions can educe phenomena like *flow separation* and *hydraulic jumps* which would require employment of higher resolutions.

This study did not intend to establish the suitability of one numerical method over other as such a pursuit would require execution of different numerical codes on the same machine, and would require that the computer science methodologies like optimization techniques, compilation options etc. followed in implementation of the said codes are in agreement. One thing that enforces superiority of spectral methods is its ability to increase effective resolution³ by increasing the polynomial order. Also Varghese, Frankel, and Fischer¹ employed higher mesh density near the walls and the stenosis throat unlike the even mesh employed by me in this study. Local grid refinement^{9,10} is implemented and validated in *Musubi* framework and an accurate gauge on resolution/memory requirements can be accomplished only by exploring those techniques. What seems obvious at this point is that very low resolutions would suffice for an accurate simulation of hydrodynamics in the post stenotic regions beyond $x > 10D$, where the flow relaminarized and did not exhibit much spatial and temporal variations.

A mentionable aspect of this study in particular, and LBM simulated flows in general is the initial transients. I had to discard the initial 2 cycles from the analysis as they contain some spurious oscillations before converging to a physically mean-

ingful outcome. Also, my specific focus on the re-laminarization of flow during acceleration and the errors in LBM that scale of the order of Ma^2 would advocate the role of a proper tuning of the LBM parameters, which can lead to *specious* density fluctuations in the flow if not well-tuned^{7,8}, and are termed as compressibility errors. For these limitations of LBM, it has generally been considered unsuitable for steady problems and its inherently transient nature makes it a suited method for time dependent flows^{7,32}. I shall not delve into details of the initial transients analysis of LBM and other methods, and a brief account for that can be found in^{7,32}.

It may be very well appreciated that the simplest off the shelf scheme of LBM reproduced an extremely complex flow with appreciable ease and efficiency. LB equations using multiple relaxation times (MRT) are intended to be more stable than the BGK as the additional relaxation times may be adjusted to suppress non-hydrodynamic modes that do not appear directly in the continuum equations, but may contribute to instabilities on the grid scale³³. Such schemes would ostensibly be useful in more complex geometries as discussed above.

514 Outlook

The present work re-validates LBM and the *Musubi* solver particularly for transitional flows. The previous benchmark is extended for the LBM, Kolmogorov scales are quantified and recommendations on the choice of spatial and temporal resolutions in simulations have emerged from the study, which have implications particularly for the simulation of physiological flows of transitional nature in complex anatomical geometries.

The aspect of the onset of transition in this stenosis in particular has been an aspect of many recent studies^{34,35}. The eccentricity in the present stenosis was introduced to trigger turbulent like flow as in a symmetric geometry and mesh, a perturbation is needed to cause the onset of turbulence¹⁴. The insights from this study can be used to explore the critical Reynolds number for transition in an axisymmetric stenosis and the influence of breaking of symmetry on critical Re can be identified with LBM, as

534 was done by Samuelsson, Tammisola, and Juniper³⁴
535 using NEK5000.

536 Insight into fundamental physiological questions
537 would be possible by incorporation of more phys-
538 iologically realistic models like for example Non-
539 Newtonian blood flow models and moving arterial
540 walls. Whereas the Kolmogorov micro-scales are the
541 smallest scales for turbulence in a flow – this hypoth-
542 esis might not hold for blood³⁶ as the interaction of
543 red blood cells (RBC) would obviate formation of
544 eddies down to the Kolmogorov micro-scales. LBM
545 models that are capable of modeling RBC interac-
546 tions³⁷ should be evaluated in *Musubi* for better in-
547 sight into such phenomena.

548 A final remark that can be made is that a LBM
549 simulation on the same geometry with $\sim 8 \times 10^6$ cells
550 was conducted using *Musubi* on my personal lap-
551 top, which completed one cycle in ~ 26 hours. The
552 fluctuations captured were less intense than the pre-
553 sented DNS, albeit the qualitative agreement was
554 very good. This suggests that with improving com-
555 puter architectures, one might be able to simulate
556 such problems on local computers with appreciable
557 ease in future.

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